

# Recearch & Technology

# cWorks – Corrosion Control System

Stephen Gaydos ASETSDefense Workshop 7 February 2011

maintaining the data needed, and c including suggestions for reducing	lection of information is estimated to ompleting and reviewing the collect this burden, to Washington Headqu uld be aware that notwithstanding an DMB control number.	ion of information. Send comments arters Services, Directorate for Infor	regarding this burden estimate mation Operations and Reports	or any other aspect of the property of the contract of the con	his collection of information, Highway, Suite 1204, Arlington			
1. REPORT DATE 07 FEB 2011 2. REPORT TYPE					3. DATES COVERED <b>00-00-2011 to 00-00-2011</b>			
4. TITLE AND SUBTITLE				5a. CONTRACT	NUMBER			
cWorks - Corrosion	n Control System			5b. GRANT NUM	MBER			
				5c. PROGRAM F	ELEMENT NUMBER			
6. AUTHOR(S)				5d. PROJECT NU	UMBER			
				5e. TASK NUME	BER			
				5f. WORK UNIT	NUMBER			
	ZATION NAME(S) AND AD 16,St. Louis,MO,63	` '		8. PERFORMING ORGANIZATION REPORT NUMBER				
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)					10. SPONSOR/MONITOR'S ACRONYM(S)			
				11. SPONSOR/MONITOR'S REPORT NUMBER(S)				
12. DISTRIBUTION/AVAIL Approved for publ	LABILITY STATEMENT ic release; distributi	on unlimited						
	OTES 11: Sustainable Surf ans, LA. Sponsored		_	Defense Worl	kshop, February 7 -			
14. ABSTRACT								
15. SUBJECT TERMS								
16. SECURITY CLASSIFIC		17. LIMITATION OF ABSTRACT	18. NUMBER OF PAGES	19a. NAME OF				
a. REPORT unclassified	b. ABSTRACT unclassified	c. THIS PAGE unclassified	Same as Report (SAR)	24	RESPONSIBLE PERSON			

**Report Documentation Page** 

Form Approved OMB No. 0704-0188 Engineering, Operations & Technology | Boeing Research & Technology

#### What is cWorks?

Disciplined process to select corrosion resistant systems

#### • Why do we need it?

- Current practices rely on engineering guidelines along with the expertise of Material & Process engineers
- Process entails an assessment of the effects of surface protection, direct material-to-material contact, heat, ultraviolet light, moisture, size and location of components, repeated loading, and a multitude of other contributory parameters
- Process is not systematic, nor quantitative, and lacks a common reference set of corrosion parameters

Copyright © 2010 Boeing. All rights reserved. EOT\_TE\_Sub\_Template.ppt | 2

Engineering, Operations & Technology | Boeing Research & Technology

- RELATIVE CORROSION INDEX (RCI) is calculated using the parameters that make up a material system
- What is the RCI?
  - The relative potential of the material system to corrode or experience corrosion on a scale of one to 100
  - One is the most corrosion resistant and 100 the least

# **Base Material Corrosion Ranking**

Material

Aluminum

Group

Alloy

2024

Engineering, Operations & Technology | Boeing Research & Technology

#### **Chemical Technology**

**Temper** 

Alclad

Alclad

Bare

R

53.75

60

How do the materials that we use on aircraft stack up to each other with regard to corrosion?

R = 10 (very corrosion resistant) to
R = 60 (poor corrosion resistance)
Group No. = Galvanic Ranking from
1 (Active) to 18 (Noble)

Material	Group	Alloy	Product	Temper	R
Stainless Steel	11	15-5PH			22.5
		PH13-8	Forging	H1000	25.625
				H1050	25.625
			Bar		25.625
		A286			22.5
		300 Series			19.375
Alloy Steel	6	H-11			41.25
		300M			41.25
		4340	160 ksi or below		41.25
			180 ksi and above		47.5
		HY-TUF			41.25
Nickel Alloys	16	MP159			16.25
		Inconel	625		16.25
			718		16.25
Titanium	17	All	All		10
Copper Alloys	15	Be-Cu			28.75
Tungsten Alloys	10	Tungsten Carbide			10

		Baic	00
	T351		60
	T3511		60
	T81		60
2090	T8E41		60
2117	T4		60
2219	T62	Alclad	53.75
		Bare	60
	T81	Alclad	53.75
		Bare	60
	T851		60
	T852		60
7050	T73		53.75
	T74		53.75
	T7451		53.75
	T74511		53.75
	T6*		53.75
	T651*		53.75
	T6511*		53.75
7075	T6 t<=0.188	Alclad	47.5
		Bare	53.75
	T6 t>0.188*	Alclad	47.5
		Bare	53.75
	T73		53.75
	T7351		53.75
	T76	Alclad	47.5
		Bare	53.75
7150	T7751		53.75
	T77511		53.75
7475	T761	Alclad	47.5
		Bare	53.75
	T7651		53.75
A357	Castings		47.5
D357			47.5
KO1			60
5052	Honeycomb	·	44.375
6061	T6		44.375

# **Surface Treatment Effectivity**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

	Fastener	Pr	imer	Topcoa	н	
	rastellel	Material	Type	Material	Type	"
				None	N/A	1.0
		None	N/A	Polyurethane	Heat resistant	0.4
		None	IN/A	Fuel Tank Coating	N/A	0.3
Organic Coatings -				Rain Erosion Tape	IV/A	0.5
			Regular	Polyurethane	Gloss	0.3
	No		Fluid Resistant	None	N/A	0.3
				Fluid Resistant Epoxy	19/7	0.4
		Epoxy		Polyurethane	N/A	0.4
				Impact Resistant	Regular	0.4
Inorganic Coatings			impact itesistant	Epoxy	Walk-on	0.4
-				<u></u>	Erosion	0.4
				None		1.0
1	Yes	None	N/A	Dry Film Lube	N/A	8.0
lacktriangledown				Hi-Kote		0.6

Material	Туре	T
	Clean/none	1.00
Aluminum	Chromic acid anodize	0.30
Aldmindin	Cadmium plate	0.60
	Nickel plate	1.10
	Clean/none	1.00
	Cadmium plate	1.20
Stainless Steel	Nickel plate	0.40
	Chrome plate	0.20
	Silver plate	0.80
	Clean/none	1.00
Titanium	Anodize	0.20
	Nickel plate	0.50
	Chrome plate	0.20

Material	I
None	1.00
Sealant	0.10
CPC	0.20
Sealant & CPC	0.02

Sealant

$$\Omega_{Base} = R \cdot I \cdot T \cdot H$$

Baseline Corrosion Potential =  $\Omega_{Rase}$ 



### **Environmental and Chemical Attack**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

- Three types of environmental chemical attack
  - Environmental Attack of the Organic Surface Finish
    - Salt Spray
    - Anti-ice
    - Blue Water
    - UV
  - Organic Surface Finish Damage
    - Sand
    - Gravel
    - Engine Exhaust
    - Maintenance
  - Environmental Attack of the Parent Material
    - Salt Spray
    - Anti-ice
    - Blue Water
    - UV

### **Environmental Attack – Organic Surface Finish**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

Coating				Ai								
Fastener	Primer Material	Primer Type	Material	Туре	UV	Salt Spray	Anti-Ice	Blue Water	Engine Exhaust	Heat	De-Ice	Leak
No	None	N/A	None	N/A	100%	100%	100%	100%	100%	100%	100%	100%
			Polyurethane	Heat resistant	100%	76%	76%	59%	100%	100%	59%	92%
			Dry film lube	N/A	50%	65%	65%	50%	40%	20%	50%	65%
			Hi-Kote	N/A	50%	76%	76%	59%	46%	24%	59%	85%
			Fuel tank coating	N/A	25%	87%	87%	67%	46%	27%	67%	85%
			Rain Erosion Tape	N/A	100%	87%	87%	67%	43%	27%	67%	79%
	Ероху	Regular (2104)	Polyurethane	Gloss	100%	100%	100%	77%	55%	27%	77%	100%
				APC	100%	87%	87%	67%	40%	31%	67%	73%
		Fluid Resistant	None	N/A	25%	100%	100%	38%	35%	15%	38%	65%
			Fluid-resistant epoxy	N/A	25%	76%	76%	100%	46%	40%	100%	85%
		Fluid Resistant - Two Coats	None	N/A	25%	100%	100%	38%	35%	15%	38%	65%
		Impact Resistant	Polyurethane	APC	100%	87%	87%	67%	40%	31%	67%	73%
			Impact-resistant	Regular	100%	100%	100%	77%	46%	27%	77%	85%
				Walk-on	100%	100%	100%	77%	46%	27%	77%	85%
				Erosion	100%	100%	100%	77%	46%	27%	77%	85%
Yes	None	N/A	None	N/A	100%	100%	100%	100%	100%	100%	100%	100%
			Dry Film Lube	N/A	100%	85%	85%	85%	87%	85%	100%	100%
			Hi-Kote	N/A	100%	100%	100%	100%	100%	100%	85%	80%

Environmental attack reduces the resistance of the organic finish – but not below that of the parent material.

Environmental attack factors negate the coating factor, H:

$$E = 1 + (H - 1) \cdot A_0 \cdot A_1 \cdots A_n \quad 0.2 \le E \le 1.0$$

# **Organic Surface Finish Damage**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

Coating Damage						
Location	Type	В				
External	None	1.00				
	Sand	0.75				
	Gravel	0.50				
	Engine Exhaust	0.75				
	Maintenance damage	0.25				
	Erosion	0.75				
Internal	None	1.00				
	Galley	0.25				
	Cargo area	0.75				
	Fuel tank	0.50				
	Maintenance damage	0.25				

0.25<u><</u>B<u><</u>1.0

Finish damage reduces the resistance of the organic finish – but not below that of the parent material.

Surface protection factor, C, takes into account the benefit of the coating system (organic & inorganic finish, sealant, & CPC) and the detriment of coating damage:

$$C = 1 - B(1 - E \cdot T \cdot I \cdot Y)$$

Where Y accounts for exposure resistance provided by an interference fit fastener:

Fasteners	Υ
Clearance Fit	1.0
Interference Fit	0.3

0.3<u><</u>Y<u><</u>1.0

#### **Environmental Attack – Parent Material**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

	D <sub>i</sub>							
		Salt		Blue	Engine			
Material	UV	Spray	Anti-Ice	Water	Exhaust	Heat	De-Ice	Leak
Aluminum	1.00	1.43	1.43	2.00	1.21	2.00	2.00	1.21
Stainless Steel	1.00	1.07	1.07	1.29	1.04	1.07	1.29	1.07
Alloy Steel	1.00	1.21	1.21	2.00	1.21	1.14	2.00	1.21
Nickel Alloys	1.00	1.03	1.03	1.14	1.04	1.00	1.14	1.03
Titanium	1.00	1.00	1.00	1.14	1.04	1.00	1.14	1.00
Copper Alloys	1.00	1.10	1.10	1.29	1.21	1.14	1.29	1.10
Tungsten Alloys	1.00	1.04	1.04	1.14	1.04	1.00	1.14	1.04
Plastics	1.43	1.01	1.01	1.14	1.21	2.00	1.14	1.01
Composite Prepreg	1.43	1.03	1.03	1.03	1.03	1.03	1.03	1.03
Composite Core	1.43	1.03	1.03	1.03	1.03	1.03	1.03	1.03

Chemical attack factors combine per the following:

$$D = D_1 + (D_2 - 1)/2 \cdots (D_n - 1)/n \quad \text{1.0$$

Resultant chemical attack factor interacts with surface protection factor, C:

$$F = C \cdot D$$

0<F<3.09

### **Moisture Intrusion**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

Location	J
Dry/Ventilated	1.00
Doors	1.17
Low spots	1.33
Bilge	1.50
Sump	1.50
Wheel Well	1.33
Trailing Edge	1.17

Drainage	K
Adequate	1.00
Inadequate	1.17

Moisture Intrusion Factors

Impact potential due to moisture:

$$M = J \cdot K$$
 1.0 $\leq$ M $\leq$ 1.76

- The total moisture intrusion risk is dependant on:
  - Location
  - Whether adequate drain paths exist

### **Beneficial Environments**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

Environmental Factors		
Condition	$N_{j}$	
None	1.0	
Pressurized	0.6	
Protected	0.5	
Internal	0.7	
Fuel	0.4	
Oil	0.3	

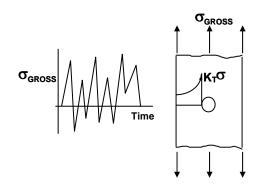
Beneficial Environment Impact

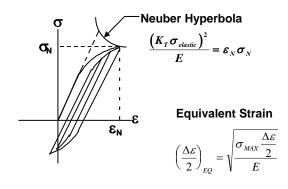
Impact potential due to beneficial environment:

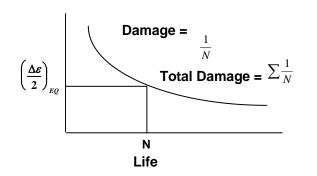
$$N = N_1 \cdot N_2 \cdots N_n \quad 0.15 \le N \le 1.0$$

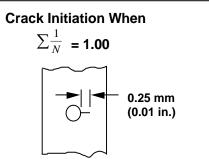
# Spectrum Effects

Engineering, Operations & Technology | Boeing Research & Technology









Spectrum Effect		
Life (Hr)	V	
> 100,000	1.0	
50,000	1.2	
10,000	1.2	
5,000	1.4	
< 1,000	1.5	

1.0<V<1.5

Fatigue Life Prediction Algorithm

# **Galvanic Activity**

Engineering, Operations & Technology | Boeing Research & Technology

#### Galvanic activity is a function of rank difference\*:

$$\Delta = \left| G_{Mat-A} - G_{Mat-B} \right|$$

\*Based on the inorganic surface treatment between two interface materials

		Galvanic Rank G		
Material	Surface Treatment	Bare	With Surface Treatment	
	Clean/none		4	
	Chromic acid anodize		4	
Aluminum	Cadmium plate 4		3	
	Nickel plate		16	
	Silver plate		18	
	Clean/none		11	
	Cadmium plate		3	
Stainless Steel	Electroless nickel	11	16	
	Chrome plate		10	
	Silver plate		18	
	Clean/none		6	
	Cadmium plate		3	
Alloy Steel	Nickel plate	6	16	
	Chrome plate		10	
	Silver plate		18	
Titanium	Clean/none		17	
	Anodize	17	17	
	Nickel plate	"	16	
	Chrome plate		10	

### **Galvanic Action**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

#### Rank Impact

Rank ∆	Р
0	1.00
6	1.22
12	1.56
18	2.00

#### **Small Anode Effect**

Rank ∆	S
0	1.00
4	1.00
8	1.12
14	1.24
18	1.36

# Organic Surface Finish Thickness Effect

Thickness	Q
0	1.00
<u>≥</u> 6	0.25

Interface corrosion potential:

$$G = P \cdot S \cdot Q$$
 0.25  $\leq$  G $\leq$  2.72

- Interface corrosion potential (risk) depends on:
  - Material potential for individual components
  - Galvanic interaction between the two materials

#### **cWorks RCI Calculation**

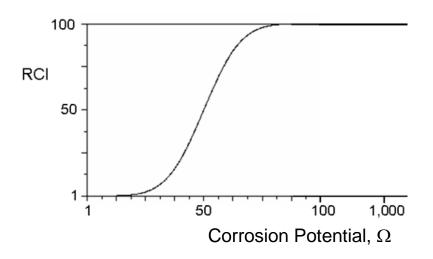
Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

 Material resistance, R, is modified based on factors for surface treatment, damage potential, moisture (location, drainage, fastener fit), galvanic action, and the environment

$$\Omega_{Mat} = R \cdot N \cdot f(F, V, M, G) \quad 1 \le \Omega_{Mat} \le 407$$

• Cumulative probability density function is used to convert the material's corrosion potential score,  $\Omega$ , to a relative corrosion index, RCI 1<a href="#">RCI</a>100



#### What can we do with cWorks?

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

- On existing aircraft: corrosion-based inspection intervals can be determined and inserted into existing maintenance plan
- In the design phase: mitigate corrosion risk items before they occur

# **cWorks Used for Inspection Intervals**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

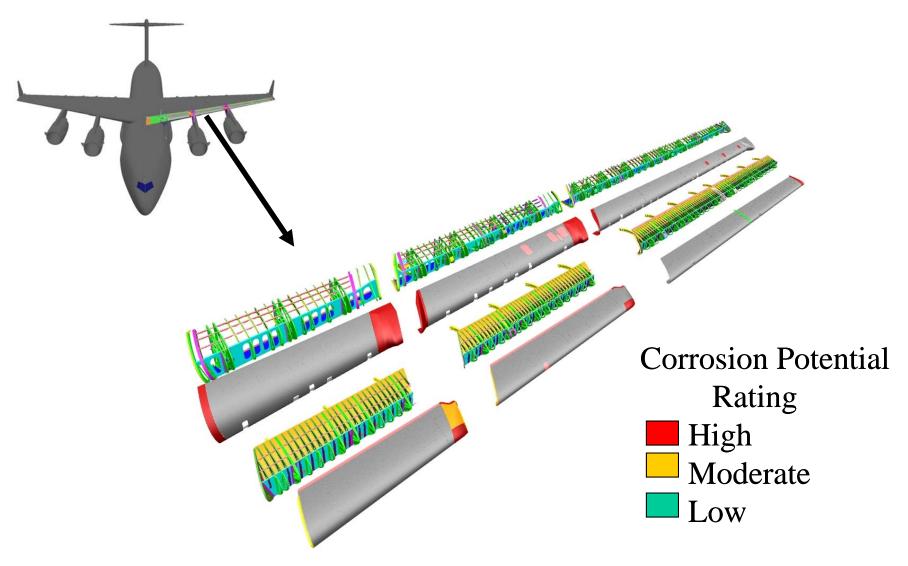
- RCI is a relative indicator of corrosion potential only
- Implement subjective relationship between RCI and inspection interval

			In-Service	
RCI	Flag	Design	Relative Inspection Interval	Inspection Interval
75 – 100		Must fix it		240 days
50 – 74		Suggest fix it		720 days
30 - 49		Marginally okay		Five years
15 - 29		Okay		Half-lifetime
0 - 14		Very good		No inspection

# cWorks Used for Design

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 



### **Current Status**

Engineering, Operations & Technology | Boeing Research & Technology

#### Current Status

- Stand-alone PC-based software program developed
- Patent Issued –

#### **United States Patent**

Meyer et al.

(10) Patent No.: US 7,704,371 B2 (45) Date of Patent: Apr. 27, 2010

#### CORROSION IDENTIFICATION AND MANAGEMENT SYSTEM

Inventors: Eric S. Meyer, Chesterfield, MO (US); Jeffrey S. Sermersheim, St. Charles, MO (US); Stephen P. Gaydos, St. Louis, MO (US); Ko-Wei Liu, Seal Beach, CA (US)

Assignee: The Boeing Company, Chicago, IL

Notice: Subject to any disclaimer, the term of this patent is extended or adjusted under 35

U.S.C. 154(b) by 806 days.

Appl. No.: 11/609,179

Filed: Dec. 11, 2006

Prior Publication Data

US 2008/0126033 A1 May 29, 2008

#### Related U.S. Application Data

Provisional application No. 60/823,537, filed on Aug. 25, 2006.

Int. Cl. C23F 13/04 (2

C23F 13/04 (2006.01) C23F 13/22 (2006.01)

- - see application the for complete search histo

#### (56) References Cited

#### U.S. PATENT DOCUMENTS

6.19	95.624	B1*	2/2001	Woodman et al 703/7
				Bosch et al 324/700
6,8	11,681	B2*	11/2004	Dowling et al 205/725
6,8	52,539	B2*	3/2005	Fields et al 702/42
7,03	29,569	B2*	4/2006	Dowling et al 205/725

<sup>\*</sup> cited by examiner

Primary Examiner—Bruce F Bell (74) Attorney, Agent, or Firm—McNees Wallace & Nurick, LLC

#### (57) ABSTRACT

The present invention is directed to a system for identifying and managing corrosion and methods related thereto. In particular, the invention provides a system for identifying and managing the structural corrosion potential for an assembly by measuring a "Relative Corrosion Index" ("RCI").

#### 28 Claims, 8 Drawing Sheets

Engineering, Operations & Technology | Boeing Research & Technology

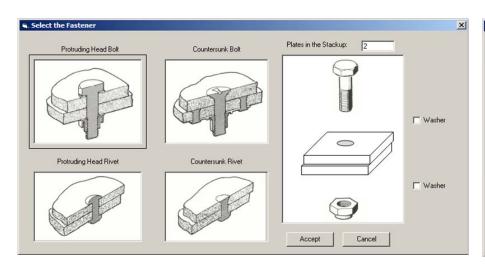
#### CWorks v2.0 Enhancements:

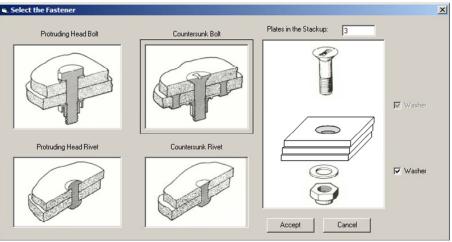
- Corrosion Ranking for Air Force Bases Added
  - Dover, Charleston, McChord, Travis, Altus, McGuire, Elmendorf, Jackson, March, Edwards, Hickam, and Foreign Bases
- Composite materials added
- Fastener joint designs added (cWorks Fastened Joint Wizard)
- User Friendly Software

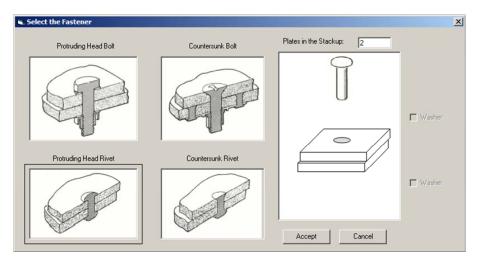
### **cWorks Fastened Joint Examples**

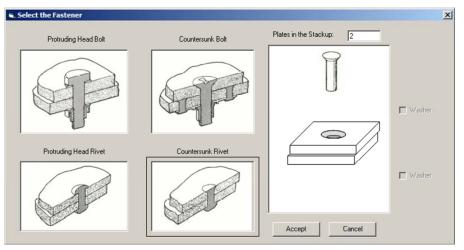
Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 







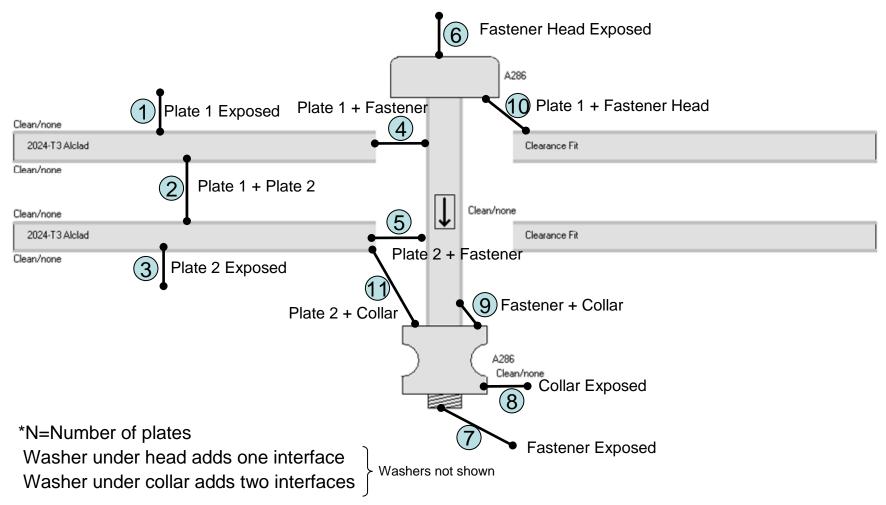


#### **cWorks Fastened Joint Wizard**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

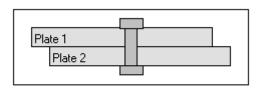
#### Wizard creates (7+2N)\* interfaces



### **cWorks Demo**

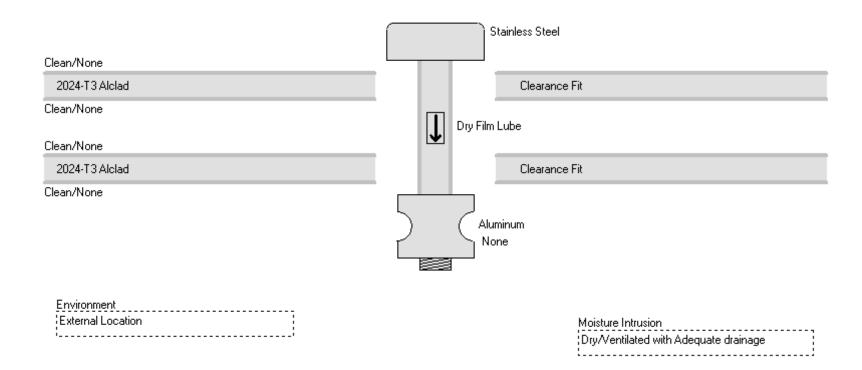
#### Engineering, Operations & Technology | Boeing Research & Technology

#### **Chemical Technology**



Moisture Intrusion Environment

Dry/Ventilated with Adequate drainage External Location



### **Questions?**

Engineering, Operations & Technology | Boeing Research & Technology

**Chemical Technology** 

